# CENTRAL INSTITUTE OF AVIATION MOTORS

# ELECTRIC DEMONSTRATION SYSTEMS

# OF THE GAS-TURBINE ENGINE FOR THE MORE ELECTRIC AIRCRAFT

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□ Main features of electric engine

□ CIAM activities on electric engine

Demonstration system of electrically driven

automatic control

Demonstration system of oil supply system with

electrically driven pumps

## **IMPORTANT AREAS OF ENGINE ELECTRIFICATION**

- Engine working process
- □ Automatic control and fuel supply systems
- Oil system
- □ Starting system
- Electric power generation system
- □ Thrust generation system

#### **LIVATION ELECTRIC ENGINE FOR THE MORE ELECTRIC** AIRCRAFT

## **Current Technology Engine**





Gearbox

Gilder Without gearbox

Electric drives

Power drives by fuel

Mechanical bearings with

oil system

Air bleed for aircraft systems

Magnetic bearings (without

**Electric Engine** 

oil system)

Without air bleed

**Embedded** starter/generator

### **UV4** OF ENGINE ELECTRIC DEVELOPMENT

- Optimization of air-gas channel for starter/generator arrangement
- **□** Engine performance optimization at reduced air bleed
- □ Distributed control system with electric drives:
  - Iow weight electric-driven actuators
  - smart sensors and actuators
  - multistage centrifugal pumps with high efficiency
- □ Oil system with electrically driven pumps

## Magnetic bearings for rotors

# CIAM ACTIVITIES ON ELECTRIC ENGINE

- □ Conception of electric engine
- Demonstration system of electrically driven automatic control
- Fuel system with electrically driven pumps
- Oil demonstration system with electrically driven pumps

 Engine test bed for testing of demonstration systems

# ine $(n_2 = 17500 \text{ rpm}, T = 17 \text{ kN})$



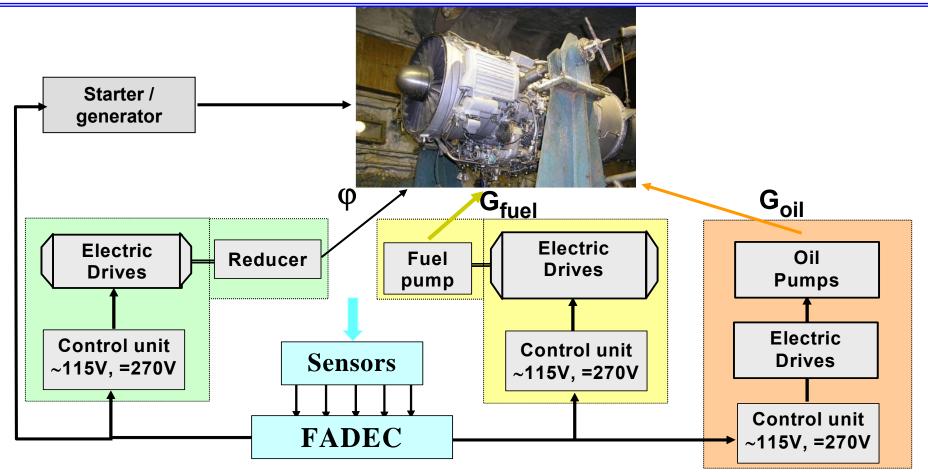
**Engine-demonstrator** 

Tests of automatic control system and oil system with electrically driven pumps



- Demonstration of electric technologies abilities
- □ Selection of actuators
- Development of a digital control system for electric drives
- Selection of control laws for electric drives management

## **DEMONSTRATION SYSTEM SCHEMATIC**



□ Sensors: rpm ( $n_2$ ,  $n_1$ ), air pressure at the compressor exit( $P_2$ ), fuel flow ( $G_{fuel}$ ), etc.

□ control units of actuators (electric drives of the pump and compressor IGV)

electromagnetic valves for control of air bleed

#### System affects fuel flow, IGV position and air bleed valves



**P**<sub>max</sub>

70 bars

Electric drive of fuel pump		IGV electric drive		Digital controller		
					2 DAC board	
N N <sub>max</sub> G <sub>Fmax</sub> P <sub>in</sub>	3 kW 12000 rpm 1000 kg/h 3 – 5 bars	N <sub>max</sub> 120	14 kW 000 rpm 0 kg	CPU speed RAM CPU clock rat Number of inp		2.2 GHz 226 MB 33 ms 10

25 mm/s

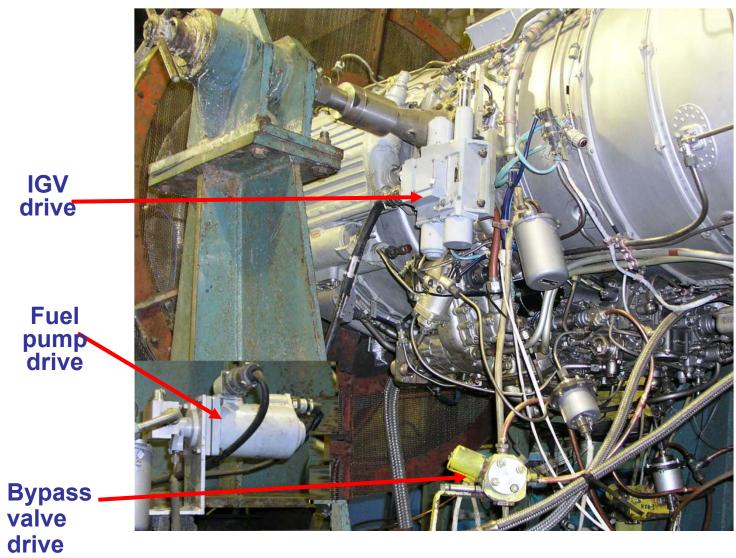
ICAS Biennial Workshop – 2013 "The More Electrical Aircraft : Achievements & perspectives for the future" Cape Town, South Africa, 2 September 2013

 $V_{max}$ 

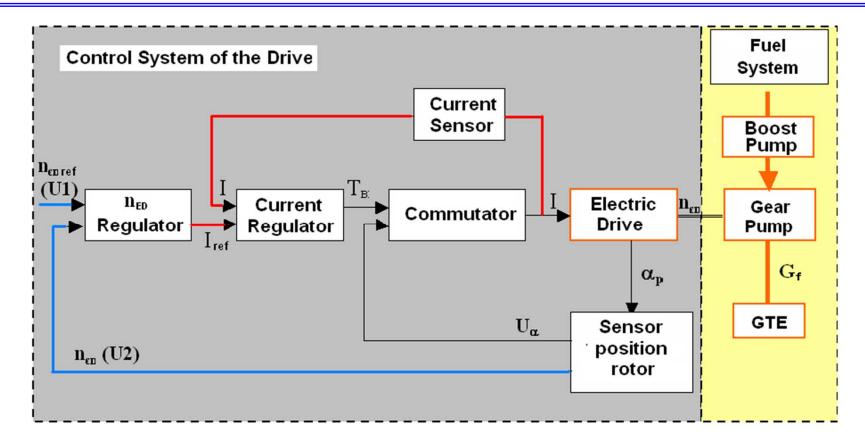
Number of output channels 12



## INSTALLATION OF UNITS ON THE ENGINE-DEMONSTRATOR



## **UNATION** DRIVE CONTROL SYSTEM SCHEMATIC



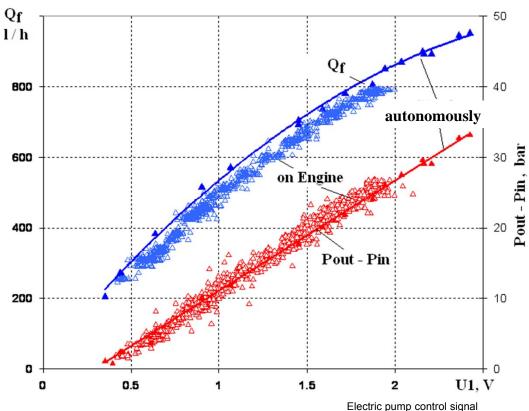
#### 2 operating modes

Control by rotational speed of electric motor rotor

#### Control by torque on electric motor rotor

# **LIVATION STATIC CHARACTERISTICS OF FUEL SYSTEM**

#### Experimental fuel flow characteristic of fuel system



- Control by rpm blue
- Control by current (torque of electric motor shaft) – red

Installed electric drive system

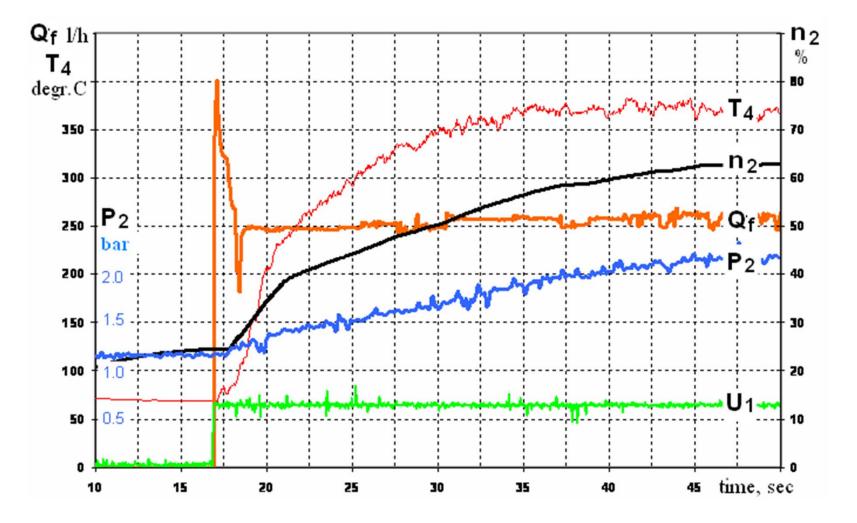
- At starting, acceleration, advanced acceleration, deceleration and stopping

Dynamic performance at control by rpm and by current are rather different

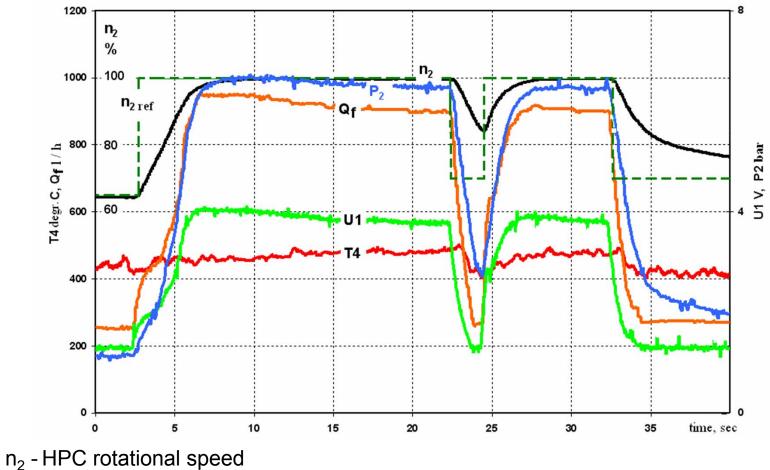
Transition time from MIN to MAX at control be rpm -0.2s (2 times less than at control by current)

control by rpm is more preferable





## **ENGINE ACCELERATION, ADVANCED** ACCELERATION AND DECELERATION



- $\bar{Q_f}$  fuel flow
- U1 control signal to the pump electric drive
- T4 turbine exit temperature

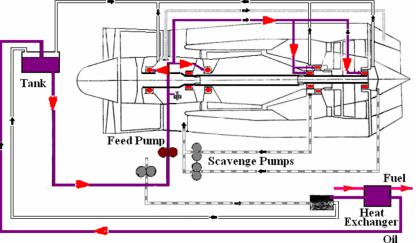


- Engine control is operable at steady and transient conditions (start, acceleration, advanced acceleration, deceleration, stopping)
- □ IGV electric drive:
  - transition time 0.5 sec
- □ Electro pneumatic bypass valve:

## opening / closing time - 0.2...0.3 sec

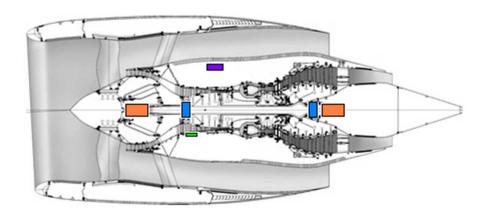
#### OIL DEMOSTRATION SYSTEM WITH ELECTRIC DRIVES

#### **Current Technology Engine**



- Oil pumps drive by HP rotor
- Breather drive by HP rotor
- System for oil heating at the engine start

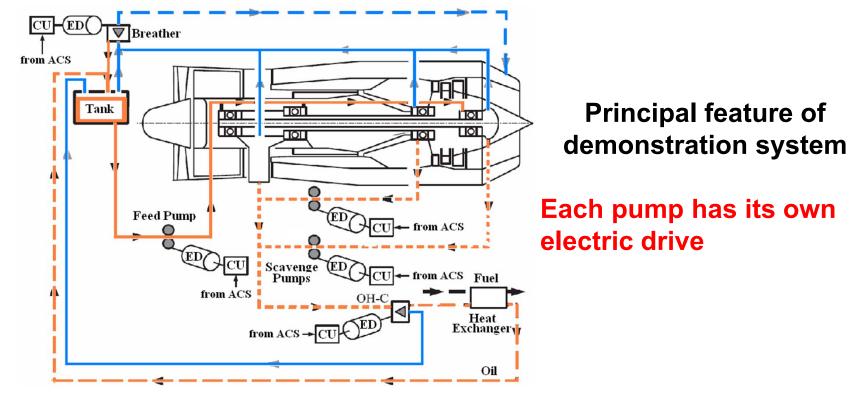
#### **Electric Engine**



- Independent electric drive feed and scavenge pumps
- > Electric drive for breather
- Electric drive for evacuation of oil from cavities



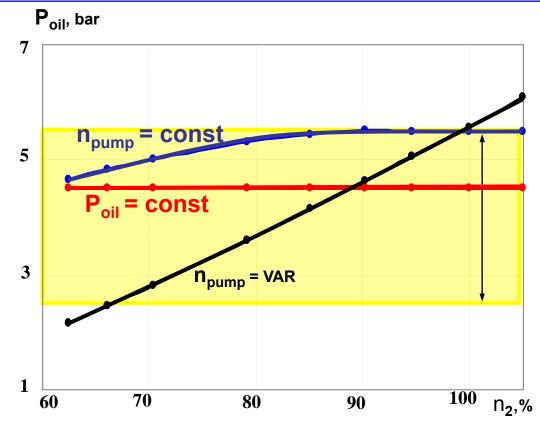
### OIL DEMONSTRATION SYSTEM WITH ELECTRICALLY DRIVEN PUMPS



## Efficiency:

- Improvement of oiling quality (increase of life time)
- Decrease of fuel heating
- Weight reduction
- Engine starting improvement at low ambient temperature (t < -30°C)</p>

#### SELECTION OF CONTROL LAWS FOR ELECTRICALLY DRIVEN PUMPS OF OIL SYSTEM



#### Laws of oil supply

- **n**<sub>pump</sub> = **const** constant rotation speed of the feed pump
- $P_{oil} = const constant$  pressure behind the feed pump (with pressure controller)
- **n**<sub>pump</sub> = **var** drive of pumps by gear box (conventional system)
- 1 and 2 laws best performance relating to oil heating

#### Preferable law n<sub>pump</sub> = const (without pressure controller)



